#### **CVL 756**

# PLASTIC ANALYSIS

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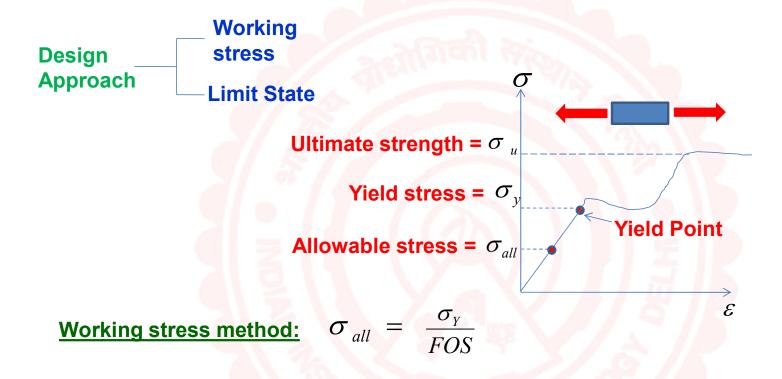
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# **ENGINEERING DESIGN APPROACH**



- -Stress is restricted to  $\sigma_{all}$  under working loads. No load factor.
- -Elastic analysis of structure is considered adequate

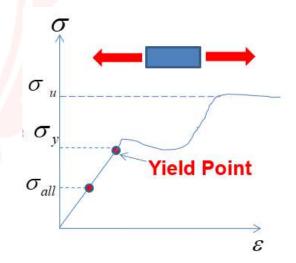
#### LIMIT STATE METHOD

- 1. Design the structure for limit state of collapse
- 2. Check for limit state of <u>serviceability</u>.
- 3. Partial factors of safety for both loads as well as stresses

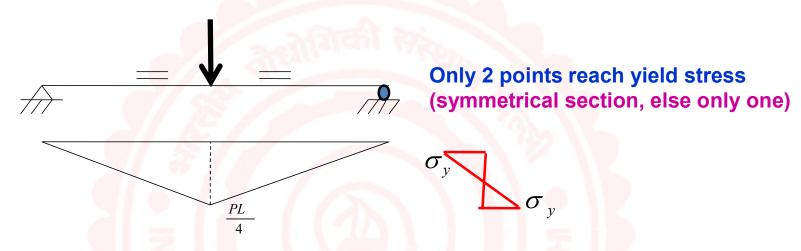
#### Conventional analysis approach for limit state design

- 1. Carry out linear elastic analysis (unfactored load values)
- 2. For limit state values of axial force, shear force and bending moment we simply multiply by load factor.
- 3. Section design, we follow rigorous non-linear computations.

# This makes the process somewhat contradictory in nature.. HOW?

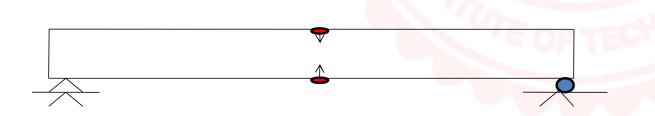


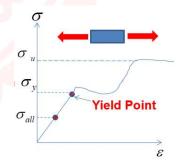
# OTHER FLAWS WITH CONVENTIONAL ANALYSIS APPROACH



Corresponding moment is called as Yield Moment M<sub>v</sub>

But the structure has not yet reached collapse state, can still carry further load





## WHY PLASTIC ANALYSIS?

The structure deemed failed at Yield Moment still has capacity to sustain higher loads and bending moment.

Actual failure values  $P_{ult}$ ,  $M_{ult}$  are much higher than

 $M_{y}$   $P_{y}$ 

By plastic analysis, we can get

Actual Load Factor 
$$\lambda = \frac{P_{ult}}{P_{v}}$$

Correspond to yielding at two points only.

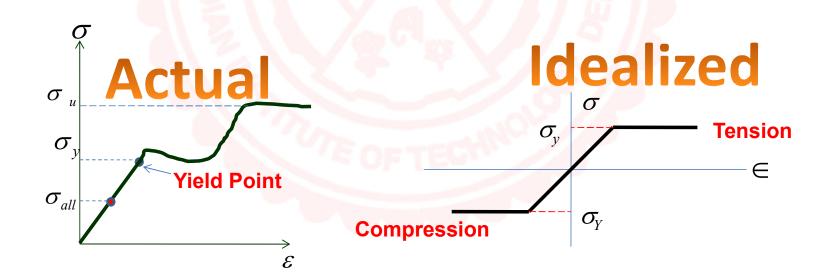
Plastic Analysis basically extends the Limit state approach (so far restricted to design aspect only) to <u>load analysis</u>

In Plastic Analysis, we take into account the actual behaviour of structure beyond the yield point.

Actual behavior of structure @ collapse (rather than yield point) is realistically taken into account.

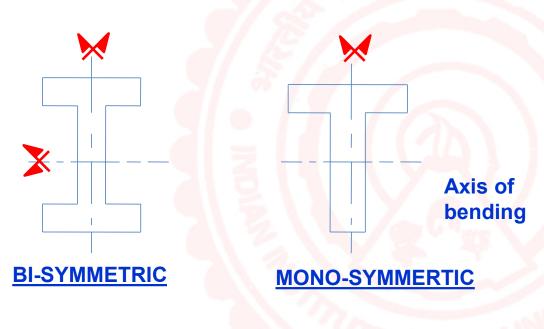
## **PLASTIC ANALYSIS: ASSUMPTIONS**

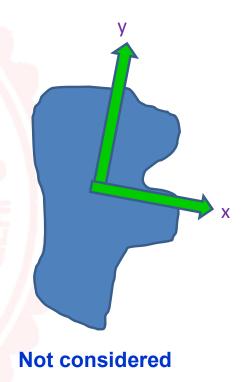
- 1) Hookes's law of elaticity holds until yield point Stress – strain curve is linear till yield point.
- 2) Yield stress and Young's modulus have same values for tension and compression.
- 3) Stress-strain curve is same in tension and compression.
- 4) Strain hardening is ignored



# **ASSUMPTIONS (Contd.).....**

5) Section has at least one line of symmetry.

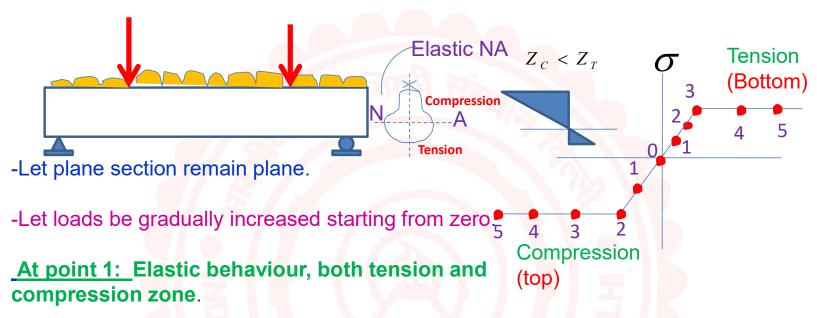


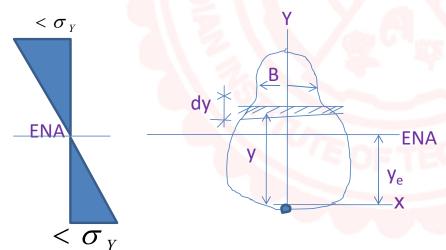


Why....???

So that the plane of loading and the plane of bending coincide with the plane of symmetry....else formulations won't apply

#### STEP-BY-STEP ANALYIS TILL COLLAPSE





C-T = No net axial force.

$$\int \sigma b dy = o$$

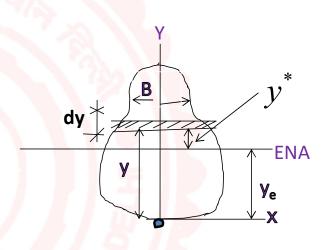
$$\sigma = \frac{M}{I} y^* = \frac{M}{I} (y - y_e)$$

$$\int \sigma b dy = 0$$

$$\int_{A} \frac{M}{I} (y - y_e) b dy = 0$$

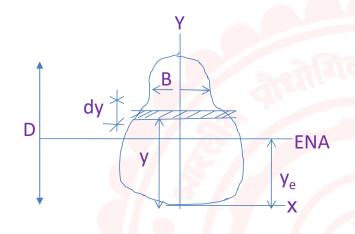
$$\int y \, b dy \, - y_e \int b dy \, = 0$$

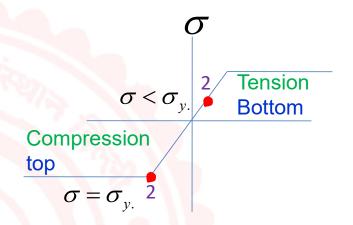
$$y_e = \frac{1}{A} \int y \ dA$$



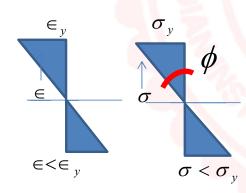
A= Total are of section.

#### **INCREASE LOADS TO REACH POINT<2>**





Yield point is reached for compression.



Yield condition arrived!

(one point only, @ compression)

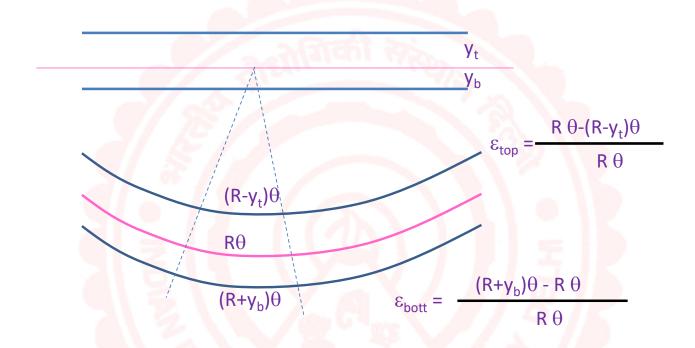
Comp. Tension

Curvature = 
$$\tan \phi = \frac{\varepsilon_{top} + \varepsilon_{bott}}{D}$$
 Absolute values

Depth of section

$$\sqrt{\phi} = \left(\frac{\epsilon_b - \epsilon_t}{D}\right)$$
 With sign convention

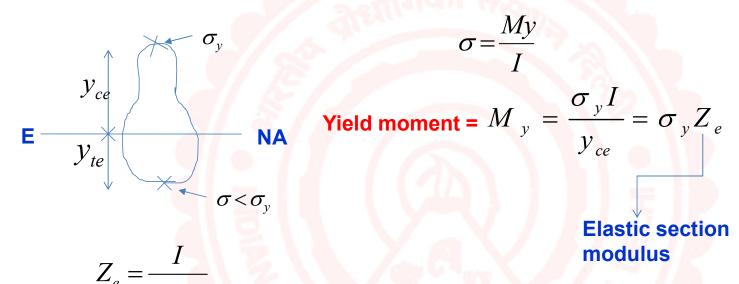
+ive value implies concave upwards



$$\phi = \frac{\varepsilon_{top} + \varepsilon_{bott}}{D}$$

## **POINT<2**> contd......

Hence moment acting an the section = YIELD MOMENT =  $M_{\nu}$ 



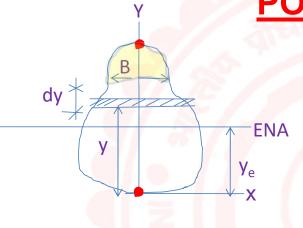
 $y_{e,\max}$ 

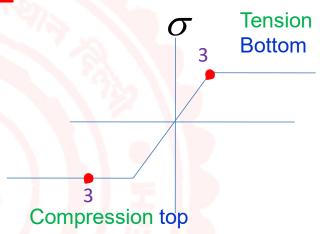
Larger of the two extreme fibre distances from ENA

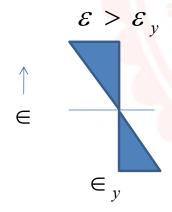
 $^{ o}$  Distance from ENA, so here equal to  $y_{ce}$ 

# INCREASE LOAD FURTHER YIELD POINT ARRIVES AT BOTTOM (TENSION)

POINT <3>







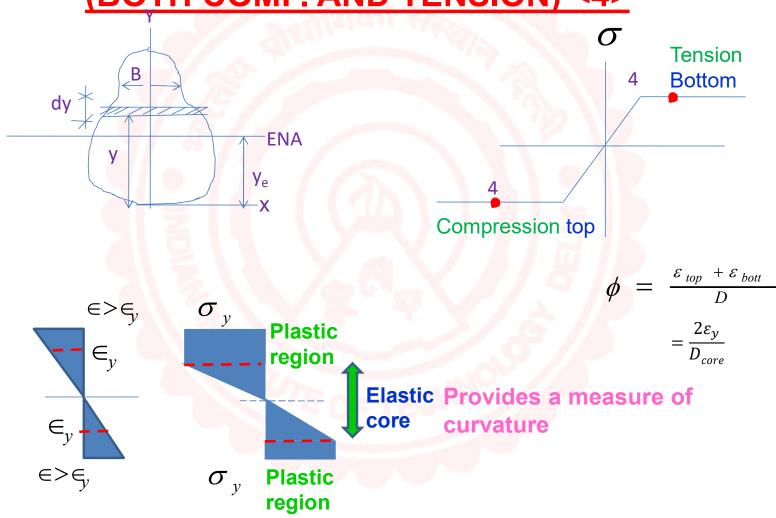


 $\sigma_{v}$ 

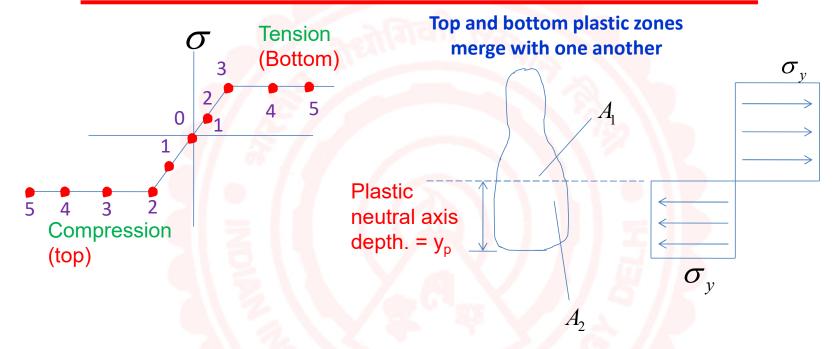
No Longer elastic neutral axis (ENA) Nor fully plastic neutral axis (PNA)

Stress no longer proportional to y

# INCREASE LOAD FURTHER SO AS TO REACH BEYOND YIELD POINT (BOTH COMP. AND TENSION) <4>



# LOAD THE STRUCTURE SUCH THAT ELASTIC CORE VANISHES POINT <5>



'NA' defined by C = T

New neutral axis is called as Plastic neutral axis (PNA)

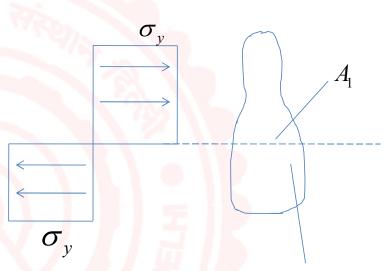
# **FULLY PLASTIC SECTION.**



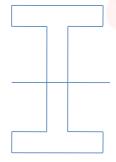
$$C = T$$

$$\sigma_{y} A_{1} = \sigma_{y} A_{2}$$

$$A_1 = A_2$$



**PNA** bifurcates the section into 2 equal areas



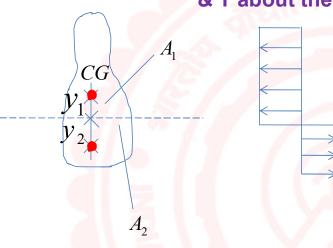
**ENA = PNA for Doubly symmetric sections** 

ENA ≠ PNA for mono symmetric sections (symmetric about y axis only)

# PLASTIC MOMENT CAPACITY AND SECTION MODULUS

 $A_1 = A_2$ 

Can be obtained by taking the moment of C & T about the PNA



$$M_P = \sigma_y Z_P$$

$$Z_P = \frac{\text{Plastic section}}{\text{modulus}} = \frac{A}{2}(y_1 + y_2)$$

Shape factor = 
$$\frac{M_p}{M_y} = f = \frac{Z_p}{Z_e}$$

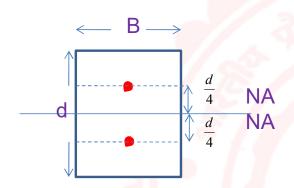
$$M_{P} = \sigma_{y} A_{1} y_{1} + \sigma_{y} A_{2} y_{2}$$

$$A_{1} = A_{2} = \frac{A}{2}$$

$$M_{P} = \sigma_{y} \frac{A}{2} (y_{1} + y_{2})$$

Distance between the CGs of two areas into which PNA divides the cross section.

# **EXAMPLE (1)**



$$Z_e = \frac{1}{6}bd^2$$

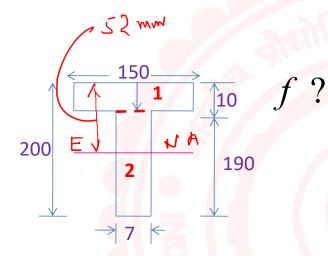
$$Z_{p} = \frac{A}{2} (y_1 + y_2)$$

$$=\frac{bd^2}{4}$$

$$f = \frac{Z_p}{Z_e} = 1.5$$

Plastic collapse moment: 50% higher than yield moment.

# **EXAMPLE** (2)



**ALL DIMENDIONS IN MM** 

Yield stress= 250MPa

$$Z_{e} ? Z_{p} ?$$

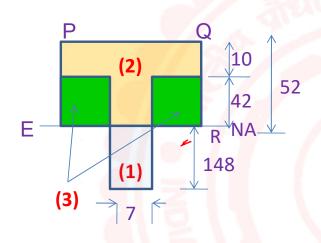
$$Y_{e} = \frac{1}{A} \int y dA$$

$$= \frac{Y_{1} A_{1} + Y_{2} A_{2}}{A_{1} + A_{2}}$$

$$= \frac{5 \times (150 \times 10) + (10 + \frac{190}{2})(190 \times 7)}{150 \times 10 + 190 \times 7}$$

$$= 52 mm$$

# **EXAMPLE** (2) contd...



$$I = \frac{1}{3}(7)(148)^3$$
(1)

$$+\frac{1}{3}(150)(52)^{3}$$
(2)

$$-\frac{1}{3}(150 - 7)(42)^{3}$$
(3)

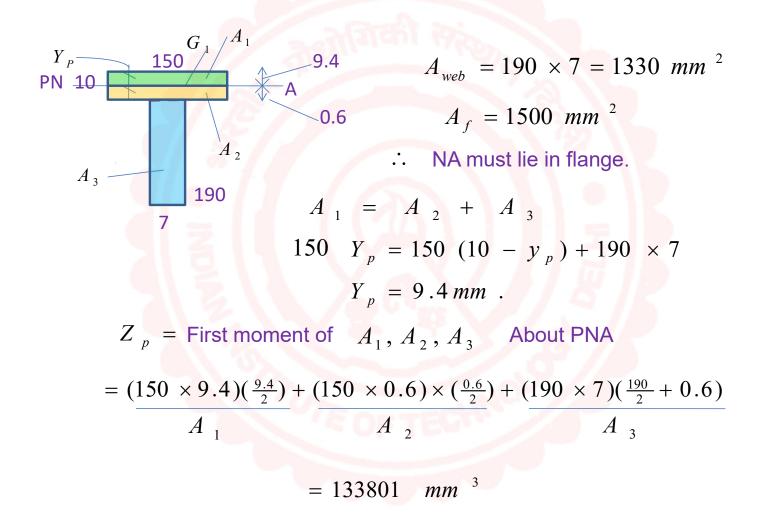
$$=11.06\times10^6 mm^4$$

$$Z_e = \frac{I}{y_{\text{max}}} = \frac{11.06 \times 10^6}{148} = 74750 \text{mm}^3$$

$$Z_{e} = \frac{I}{y_{\text{max}}} = \frac{11.06 \times 10^{6}}{148} = 74750 \text{mm}^{3} \qquad M_{y} = \sigma_{y} Z_{e} = 18.66 \times 10^{6} \text{Nmm} = 18.66 \text{kNm}$$

$$250$$

## **PLASTIC MOMENT**



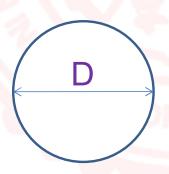
#### **M**<sub>p</sub> = Plastic moment capacity

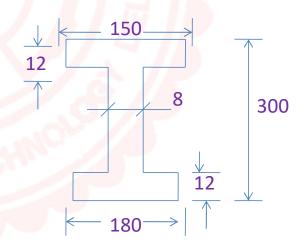
$$= \sigma_y Z_p$$
= 33.45 × 10<sup>6</sup> Nmm
$$= 33.45 kNm$$

$$f = \frac{Mp}{My} = \frac{Zp}{Ze} = \frac{33.45}{18.66} = 1.79$$

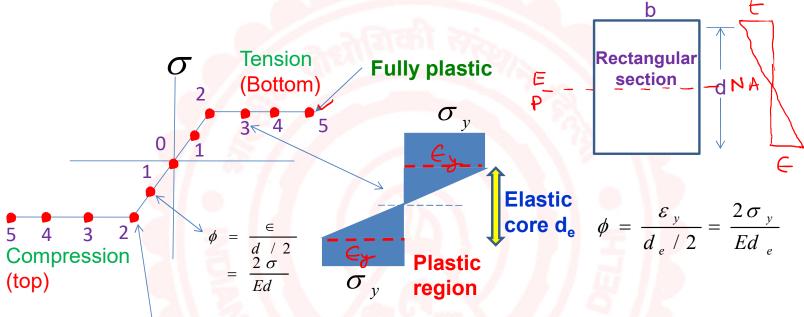
<u>H.W:</u>

Determine f





# MOMENT CURVATURE RELATION



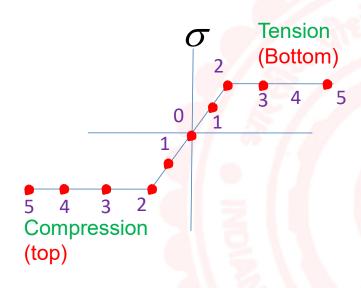
 $M_y = \sigma_y Z_e = \frac{1}{6} \sigma_y b d^2$ 

-Plastic (end) regions – will undergo strain without any increase in stress.

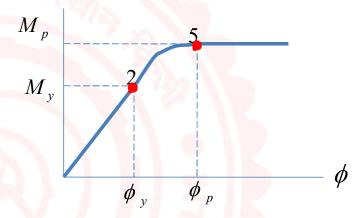
-Any moment beyond  $M_y$  will be resisted by shrinking of the elastic core .

Fully plastic condition  $d_e \rightarrow 0$   $\phi \rightarrow \infty$ 

# **MOMENT CURVATURE RELATION**



#### **Moment curvature plot**



For linear portion of curve

$$\in <\in_{y} or \sigma < \sigma_{y}$$

$$\phi_{y} = \frac{2\sigma_{y}}{Ed}$$

$$\phi_{y} = \frac{2M_{y}}{EdZ_{e}}$$

## MOMENT CURVATURE RELAT



$$M = M_{core} + M_{plastic}$$

$$M = \sigma_y Z_{ec} + \sigma_y b(\frac{d-de}{2})(\frac{d+de}{2})$$

$$\frac{1}{6}bd_e^2$$

$$M = \frac{1}{2} M_y \left[ 3 - \left( \frac{de}{d} \right)^2 \right] M_y < M < M_p$$

## MOMENT CURVATURE RELAT

$$M_y < M < M_p$$
 or

$$\phi_y < \phi < \phi_p$$

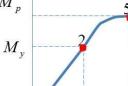
$$M = \frac{1}{2} M_y \left[ 3 - \left( \frac{de}{d} \right)^2 \right]$$

$$\phi = \frac{2\sigma_y}{Ed_e}$$

$$\phi_y = \frac{2\sigma_y}{Ed} \quad M_p$$

$$\phi = \frac{2\sigma_y}{Ed_e} \qquad \frac{\phi}{\phi_y} = \frac{d}{d_e}$$





$$M = \frac{3}{2} M_y - \frac{1}{2} M_y (\frac{\phi_y}{\phi})^2$$

$$\frac{\phi}{\phi_y} = \frac{1}{\sqrt{3 - 2\left(\frac{M}{M_y}\right)}}$$

$$M_y = \frac{1}{6}\sigma_y bd^2$$

Condition 
$$M = M_p$$

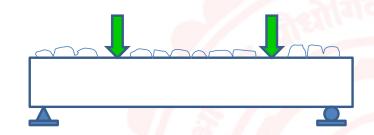
$$M_P = \frac{1}{4}\sigma_y bd^2$$

$$\frac{M_{p}}{M_{y}} = \frac{3}{2} \qquad \phi \rightarrow \infty \qquad \text{H.W. } \phi_{p} ??$$

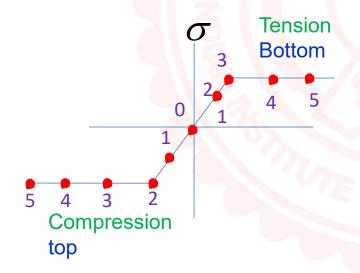
$$\phi \rightarrow \infty$$

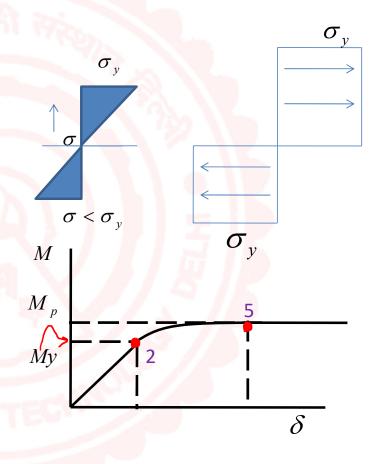
H.W. 
$$\phi_p$$
 ?'

# **MOMENT DEFLECTION RELATION**

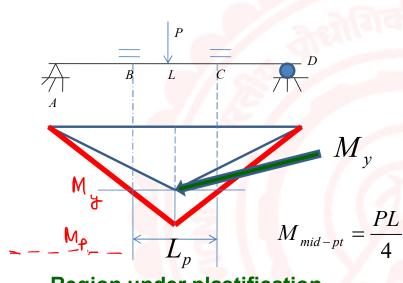


#### Assume that structure is determinate





## **SEQUENCE OF PLASTIFICATION**

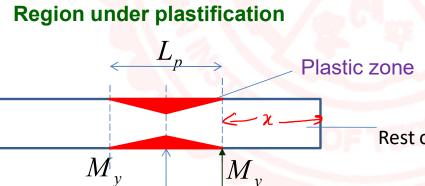




Increase load gradually.

$$P = P_{y} \qquad M = M_{y}$$

Increase load beyond P<sub>v</sub>



 $>M_{v}$ 

$$M_y < M < M_P$$

$$\frac{7}{2} \times 2 \qquad N = 1.2 My$$

$$= 1 - 4 \left(\frac{My}{P}\right)$$

Rest of the beam is elastic.

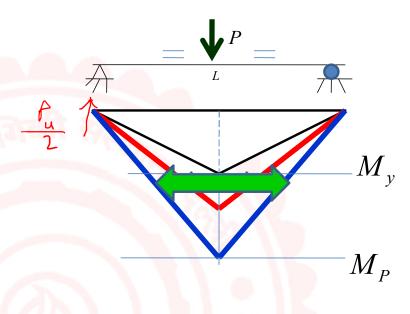
$$M = M_y = \frac{P}{2} \left( \frac{L - L_p}{2} \right)$$

#### Increase P such that

$$P=P_u$$
  $M=M_P$ 

$$M_P = \frac{P_u L}{4}$$

$$\Rightarrow P_u = \frac{4M_P}{L}$$



Limiting length of plastic zone

$$\frac{P_u}{2} \left( \frac{L - L_{po}}{2} \right) = M_y \qquad M_P = \frac{P_u L}{4}$$

$$M_P = \frac{P_u L}{4}$$

 $M_{y}$   $M_{p}$ 

 $M_{y}$ 

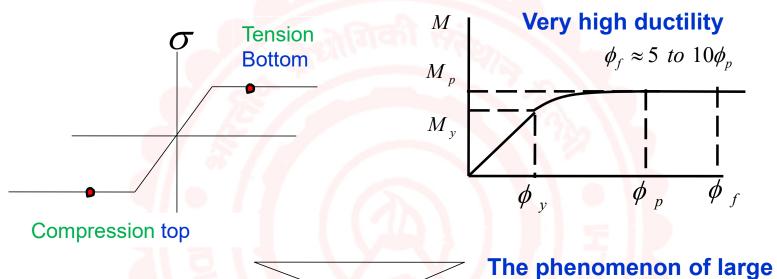
Solving the two equations

$$\frac{L_{po}}{L} = 1 - f^{-1}$$

Rectangular section: 0.33

H.W.: Shape of plastic region?

# **MOMENT CURVATURE RELATION**

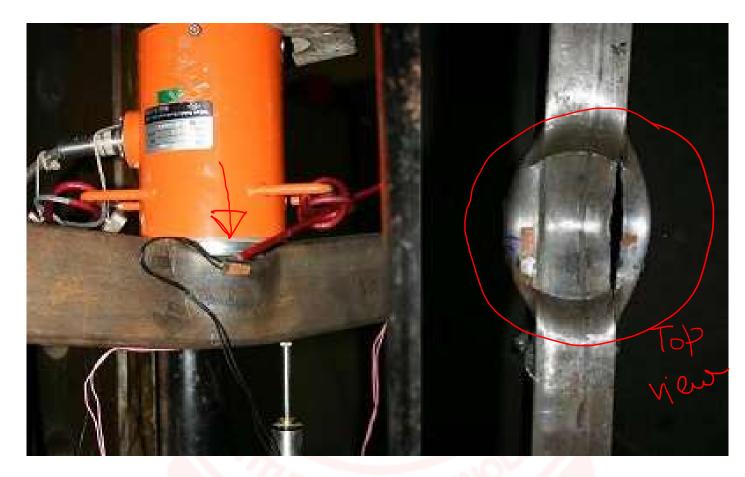


increase in curvature when load reaches P<sub>u</sub> is called as "unrestricted plastic flow"

Actual

Theoretical

Variation of curvature



TYPICAL PLASTIC HINGE

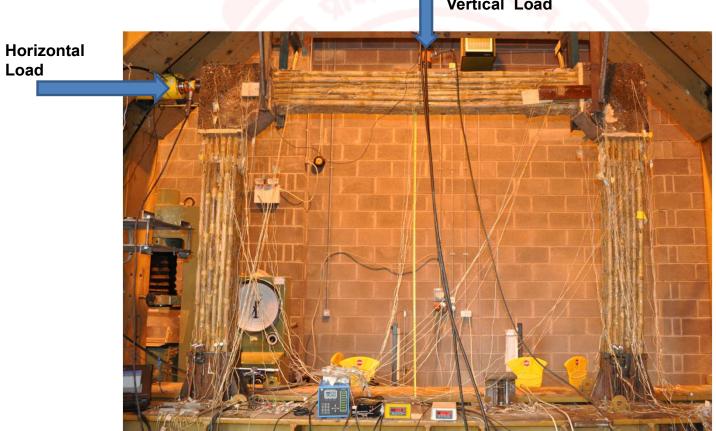




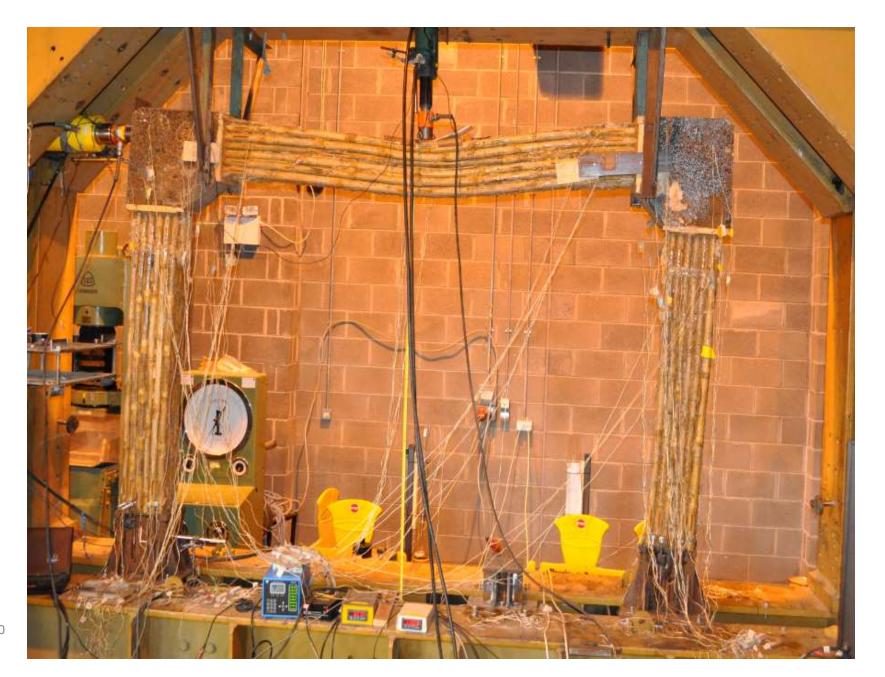


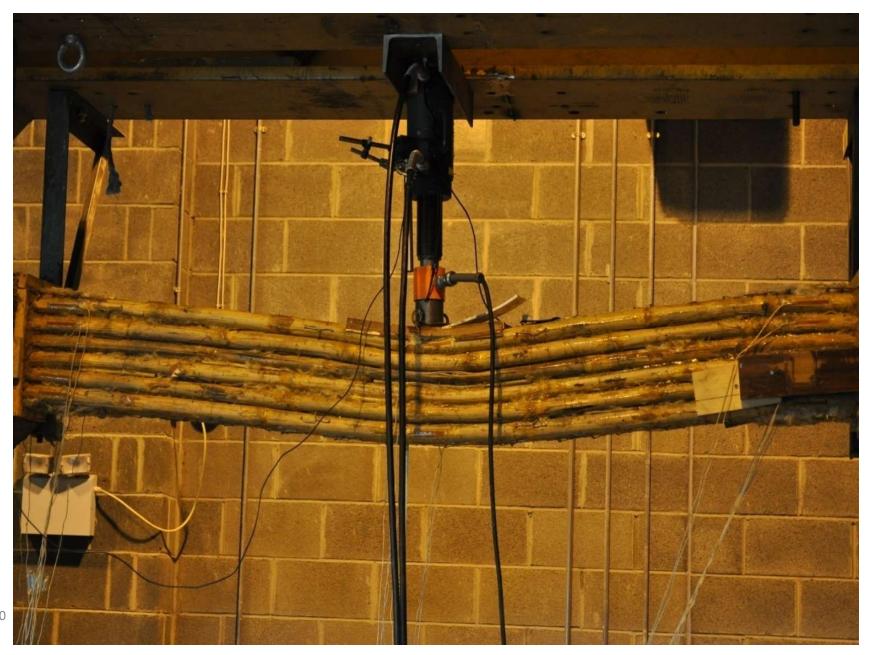
# LOAD APPLICATION

Vertical Load



11/25/2020 35





11/25/2020

# PROVISIONS OF IS 800 (2007) SEC 3.7 CLASSIFICATION OF SECTIONS

Plate elements of a cross-section may buckle locally due to compressive stresses. The local buckling can be avoided before the limit state is achieved by limiting the width to thickness ratio of each element of a cross-section subjected to compression due to axial force, moment or shear.

When plastic analysis is used, the members shall be capable of forming plastic hinges with sufficient rotation capacity (ductility) without local buckling, to enable the redistribution of bending moment required before formation of the failure mechanism.

When elastic analysis is used, the member shall be capable of developing the yield stress under compression without local buckling.

Class 1 (Plastic) — Cross-sections, which can develop plastic hinges and have the rotation capacity required for failure of the structure by formation of plastic of mechanism.

Class 2 (Compact) — Cross-sections, which can develop plastic moment of resistance, but have inadequate plastic hinge rotation capacity for formation of plastic mechanism, due to local buckling.

Class 3 (Semi-compact) — Cross-sections, in which the extreme fiber in compression can reach yield stress, but cannot develop the plastic moment of resistance, due to local buckling.

Class 4 (Slender) Cross-sections in which the elements buckle locally even before reaching yield stress.

IS 800: 2007

Table 2 Limiting Width to Thickness Ratio

(Clauses 3.7.2 and 3.7.4)

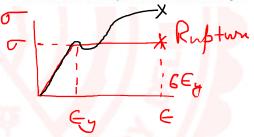
Compression Element (1)			Ratio	Class of Section			
				Class 1 Plastic	Class 2 Compact	Class 3 Semi-compact	
				(2)	(3)	(4)	(5)
Outstanding element of compression flange		Rolled section		b/t <sub>f</sub>	9.4€	10.5ε	15.7€
		Welded section		b/t <sub>f</sub>	8.4€	9.4€	13.6ε
Internal element of compression flange		Compression due to bending		b/t <sub>f</sub>	29.3€	33.5 €	42ε
		Axial compression		b/t <sub>f</sub>	Not applicable		
	Neutral axis at mid-depth		d∕t <sub>w</sub>	84€	105€	126ε	
Web of an I, H or box section		If r <sub>1</sub> is	negative:	$d/t_{\rm w}$	84€		vate Windows PC sett <b>l26.0 c</b> activate
	Generally				1+ 1,	105.0 €	1+2r2

#### 4.5.2 Requirements

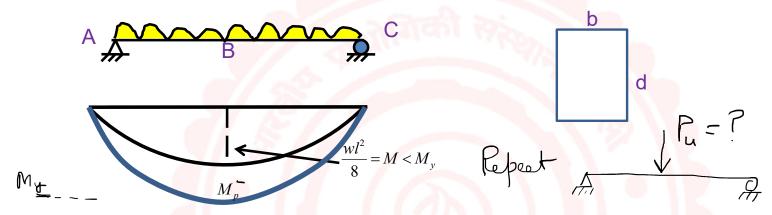
When a plastic method of analysis is used, all of the following conditions shall be satisfied, unless adequate ductility of the structure and plastic rotation capacity of its members and connections are established for the design loading conditions by other means of evaluation:

- The yield stress of the grade of the steel used shall not exceed 450 MPa.
- shall not be significantly different from those obtained for steels complying with IS 2062 or equivalent and shall be such as to ensure complete plastic moment redistribution. The stress-strain diagram shall have a plateau at the yield stress, extending for at least six times the yield strain. The ratio of the tensile strength to the yield stress specified for the grade of the steel shall not be less than 1.2. The elongation on a gauge length complying with IS 2062 shall not be than 15 percent, and the steel shall exhibit strain-hardening capability. Steels conforming to IS 2062 shall be deemed to satisfy the above requirements.
- The members used shall be hot-rolled or fabricated using hot-rolled plates and sections.
- d) The cross-section of members not containing plastic hinges should be at least that of compact section (see 3.7.2), unless the members meet the strength requirements from elastic analysis.

- e) Where plastic hinges occur in a member, the proportions of its cross-section should not exceed the limiting values for plastic section given in 3.7.2.
- f) The cross-section should be symmetrical about its axis perpendicular to the axis of the plastic hinge rotation.
- g) The members shall not be subject to impact loading, requiring fracture assessment or fluctuating loading, requiring a fatigue assessment (see Section 13).



# EX 1: SIMPLY SUPPORTED BEAM UNDER U.D.L

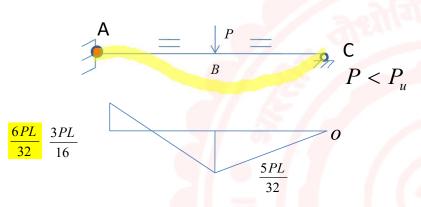


 $M_{p}^{When} = M > M_{y} \text{ but } M < M_{p}$  No change in shape of BMD continue till  $M_{p}$ 

$$But \ when \ M = M_p \qquad \begin{array}{ll} \text{Plastic hinge @ B} \\ \text{Mechanism formation} \end{array} \qquad w = w_u$$
 
$$\frac{w_u l^2}{8} = M_p = Z_p \sigma_y = \frac{1}{4} b d^{-2} \sigma_y$$
 
$$w_u = \frac{2\sigma_y b d^2}{L^2} \quad \text{= Ultimate collapse load.}$$

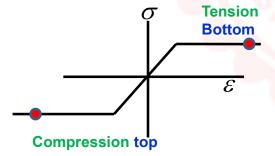
Compare with yield capacity

# **EX 2: PROPPED CANTILEVER**



# 1st Plastic hinge shall form at 'A' Collapse will not occur

Let 
$$P = P_{H1}$$
 when  $BM$  @  $A = M_p$ 



$$\frac{PL}{16}$$

$$\frac{PL}{16}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

$$\frac{P}{8}$$

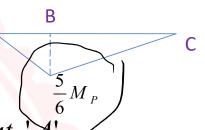
$$\frac{3P_{H1}l}{16} = M_p \qquad P_{H1} = \frac{16M_p}{3L}$$

Structure can still carry further loads

### **EX 2: PROPPED CANTILEVER**



When  $P = P_{HI}$ 



Increase "P" after formation of Plastic hinger at

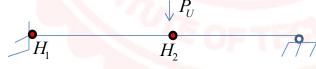
For load  $P > P_{HI} \rightarrow B.M$  cannot increase @ A

When  $P > P_{HI}$ : for incremental load

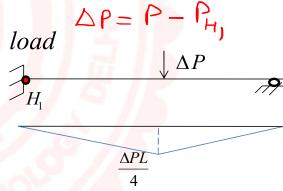
$$\Delta P = P - P_{HI}$$

When  $P = P_u, \Delta P = \Delta P_u$ 

Net BM @ 
$$B = \frac{5}{6}M_P + \frac{\Delta PL}{4}$$



$$M_p = \frac{5}{6} M_p + \frac{\Delta P_U L}{4}$$



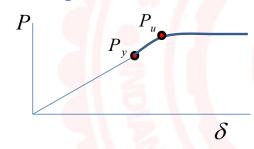
$$\Delta P_U = \frac{2}{3} \left( \frac{M_p}{L} \right)$$

$$P_u = P_{H1} + \Delta P_u$$

$$P_u = \frac{6 M_P}{L}$$

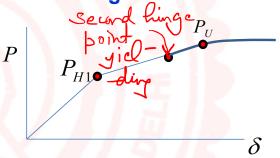
### **Determinate structures**

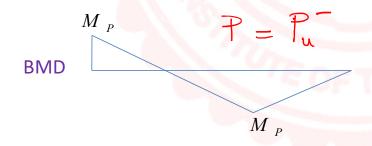
# Shape of BMD does not change till failure



### **Indeterminate structures**

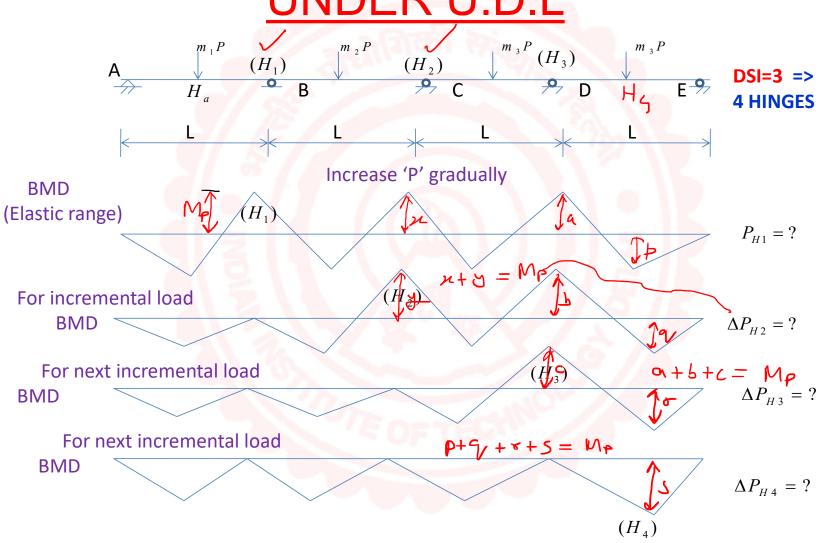
Shape of BMD changes after each hinge formation.



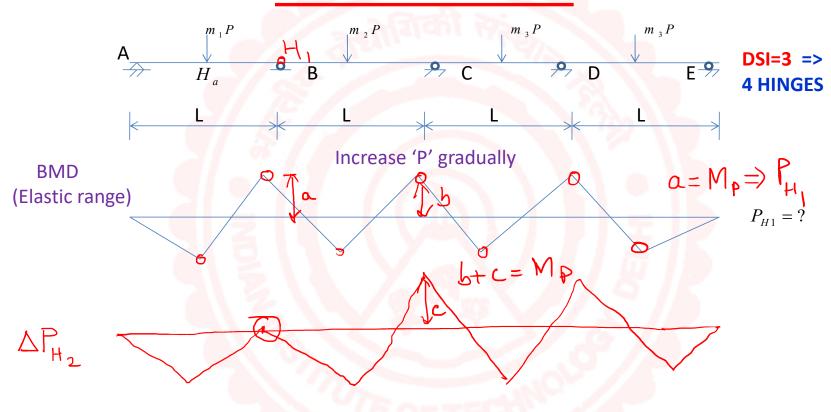


Final reactions @collapse ???

# MULTISPAN INDERTMINATE BEAM UNDER U.D.L

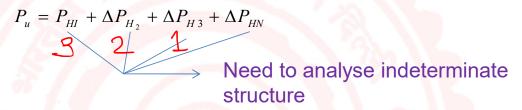


# MULTISPAN INDERTMINATE BEAM UNDER U.D.L



# INDETERMINATE STRUCTURES

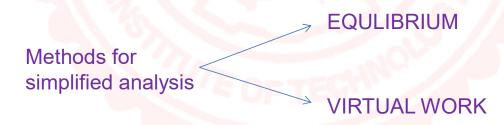
The solution process very tedious



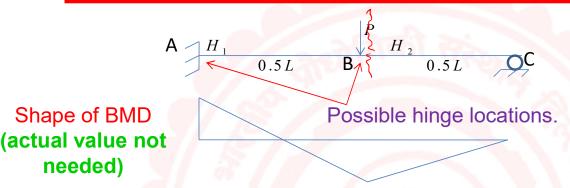
We also get to know.....

....the sequence of hinge formation.

Often, we are not interested in sequence.



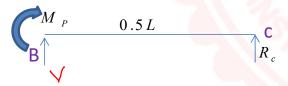
## **EQUILIBRIUM APPROACH**



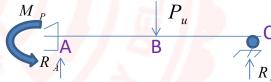
Consider equilibrium when structure is at verge of collapse

# $P = P^{-}_{u}$ Consider overall equilibrium

### Consider equilibrium of BC



Take moment about B  $R_c = \frac{2Mp}{I}$ 



Moment about A

$$M_P + (\frac{2Mp}{L})L = P_u \times 0.5L$$

Solving 
$$P_u = \frac{6M_p}{L}$$

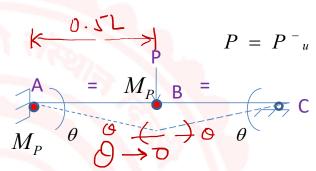
Advantage – Skip solving an indeterminate structure, which is otherwise tedious....

### VIRTUAL WORK APPROACH

#### **ASSUMPTIONS:**



- 1. Structure @ verge of collapse
- 2. Geometry of the structure not change as a result of plastification.



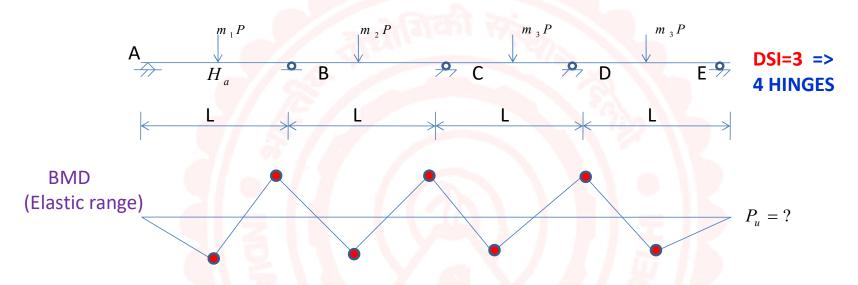
Identify hinge locations and apply small virtual (RIGID BODY) rotation  $\theta$ .

Ext. Virtual work = Internal Virtual work 
$$P_U \frac{L}{2}\theta = M_P\theta + M_P(2\theta)$$
 into

Only M<sub>p</sub> does the internal work, WHY?

Which method would you judge best out of the three methods??

# APPLICATION OF SHORT-CUT APPROACH ON COMPLICATED STRUCTURE



Problem: Four hinges till collapse, but seven possible locations

Which four of the seven possible hinges will be the governing mechanism....???

This won't be problem in the long-hand method of sequential formation of hinges...

In equilibrium or virtual work method, we rely on some important theorems..

### THEOREMS FOR PLASTIC ANALYSIS

**ASSUMPTIONS:** At the point of collapse

- (1) Loading system not affected
- (2) Geometry of structure not affected.

### **UNIQUENESS THEOREM (UT)**

Following conditions must be satisfied simultaneously at point of collapse of structure.

#### 1) **EQUILIBRIUM CONDITION**

All bending moments in equilibrium with the applied forces and reactions

#### 2) YIELD CONDITION

At all points of the structure, the bending moment  $\leq M_P$ 

3) MECHANISM CONDITION - Sufficient number of plastic hinges are formed so that mechanism condition results.....

### **UPPER BOUND THEOREM (UBT)**

If a loading can be found such that a mechanism is formed, then

$$P > P_U$$

### **LOWER BOUND THEOREM (LBT)**

If a distribution of moments can be found such that equilibrium and yield conditions are satisfied.....

...This would imply that the structure is either safe or just at the verge of collapse

$$P < P_{U}$$

If LBT and UBT are satisfied simultaneously

$$P = P_U$$

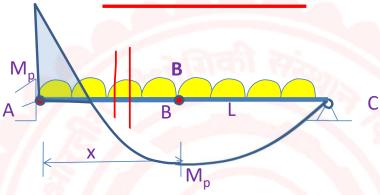
### **ANAYLYSIS PROCEDURE**

- (1) Formulate possible collapse mechanisms, based on indeterminacy. (Guesswork involved!)
- (2) According to UBT  $P_U < (P_{1u}, P_{2u}, P_{3u}, .....)$
- (3) Critical value of P<sub>u</sub>: Lowest among these values ......provided all possible mechanisms included.

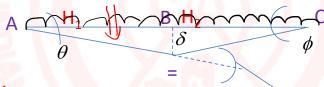
#### Possibility of missing the critical mechanism exists.

- (4) To rule out missing of critical mechanism, apply LBT (Yield and equilibrium criteria)  $P_u \ge P_{crit}$
- (5) If both LBT & UBT satisfied simultaneously imply correct collapse load  $P = P_u$

### **EXAMPLE**



Let structure be @ verge of collapse. Apply virtual displacement & @ B



External virtual work

$$\theta + \phi$$

 $\theta + \phi$  Internal virtual work

$$(wx) \frac{\delta}{2} + w(l-x) \frac{\delta}{2} = M_{P}\theta + M_{P}(\theta + \phi)$$

$$\boxed{@A}$$

Disp. of CG

$$\theta = \frac{\delta}{x} \quad \phi = \frac{\delta}{L-x}$$

Solving 
$$W = \left(\frac{2M_p}{L}\right) \left[\frac{2L-x}{(L-x)x}\right]$$

Solving  $W = \left(\frac{2M_p}{L}\right) \left[\frac{2L-x}{(L-x)x}\right]$  Infinite mechanisms depending an value of x

### **EXAMPLE (CONTD..)**

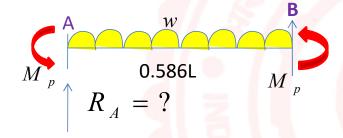
To get lowest 
$$\frac{dw}{dx} = 0 \Rightarrow x = 0.586 L$$
  
 $w_u = w_{(x=0.586 L)} = 11.65 \left(\frac{M_p}{L^2}\right)$ 

According to UBT 
$$W_u > W_{uo}$$

To ensure correct mechanism, use LBT & check <u>yield</u> & <u>equilibrium</u> condition.

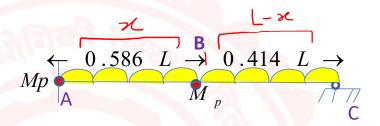
### **CHECK BY LBT**

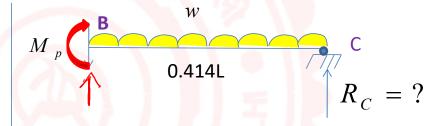
### **Equilibrium Check:**



#### Moment about 'B'

$$R_A \times 0.586 \ L = w \frac{x^2}{2} + 2 M_p$$
  
 $R_A = 0.586 \ wL$ 





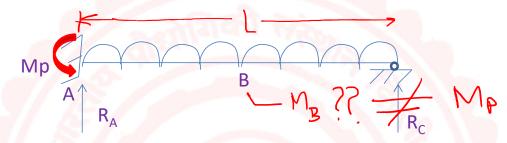
### Moment about 'B'

$$R_{C}(L-x) = M_{p} + w \frac{(L-x)^{2}}{2}$$
  
 $R_{C} = 0.4140 \ wL$ 

Check 
$$R_A + R_C = wL$$

.. Equilibrium condition satisfied

### **YIELD CHECK**



REACTIONS ON THE BASIS OF THE ABOVE BOUNDARY CONDITIONS ONLY

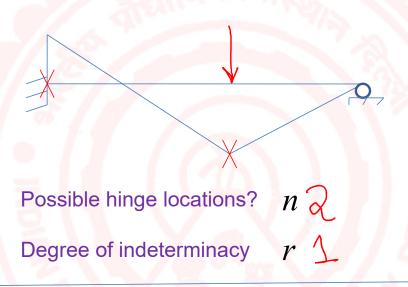
$$R_{A} = 6.83 \frac{Mp}{L} \qquad Rc = 4.83 \frac{Mp}{L}$$

$$Compute \qquad M_{B} = R_{A}x - \frac{wx^{2}}{2} - M_{P}$$

$$= ?? \qquad P$$

Alternately, we can keep  $M_{\text{A}}$  as unknown fix  $M_{\text{B}}$  to a value equal to  $M_{\text{P}}$ 

## NUMBER OF INDEPENDENT MECHANISMS



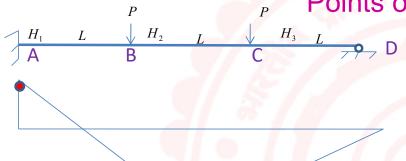
Possible **independent** mechanisms m = n - 1  $H_2 = 2 - 1$ 

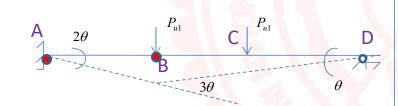
**CAUTION: THIS NUMBER IS ONLY SUGGESTIVE** 

# **EXAMPLE**

How to identify possible hinge locations ???

Points of discontinuity/ maxima of BMD





Ext virtual work = Int. Virtual work

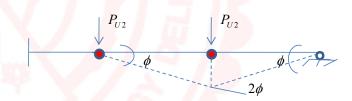
$$P_{u1}(2\theta L) + P_{u1}(L\theta) = M_{p}(2\theta) + M_{p}(3\theta)$$

$$P_{u1} = \frac{5}{3} \left( \frac{Mp}{L} \right)$$

$$m = n - n$$

$$= 3 - 1$$

$$= 2$$



Ext virtual work = Int. Virtual work

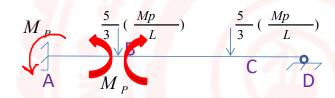
$$P_{U2}(\phi l) = M_{P}\phi + Mp (2\phi)$$

$$P_{U2} = 3(\frac{Mp}{L})$$

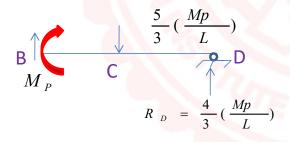
### MECH 1: EQUILIBRIUM CHECK

$$UBT \qquad \Rightarrow \qquad P_{U-1} , P_{U-2} , > P_{U}$$

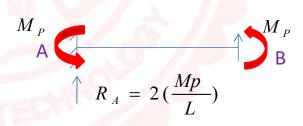
LBT – To rule out missing any mechanism. Check for equilibrium & yield.



### Consider equilibrium of BCD



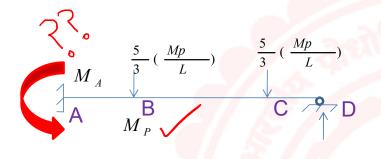
### Consider equilibrium of AB



Check : 
$$R_A + R_D = \frac{10}{3} \frac{Mp}{L} = 2 P_{U1}$$

∴ Satified

## MECH 1: YIELD CHECK

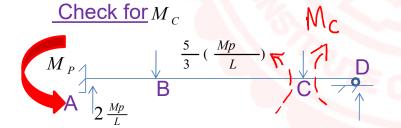


Use BCD- 
$$R_D = \frac{4}{3} \left( \frac{Mp}{L} \right)$$

Vertical equilibrium

$$R_A = 2(\frac{Mp}{L})$$

Overall equilibrium - 
$$M_A + \frac{4}{3} \left( \frac{Mp}{L} \right) \times 3L = \frac{5}{3} \left( \frac{Mp}{L} \right) \left[ L + 2L \right]$$
  
 $M_A = M_P$ 



Using CD

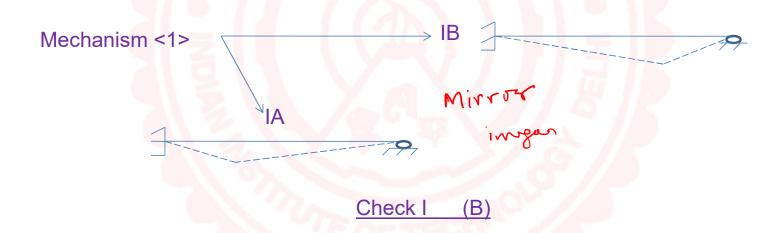
$$M_C = \frac{4}{3}(M_P)$$

Yield criteria not satisfied

=> WE MISSED OUT A CRITICAL MECHANISM

# NUMBER OF INDEPENDENT MECHANISMS

LBT- Some mechanism missed out.....



### MECHANISM I(B): MISSED OUT



External virtual work = Internal virtual work

$$P_{U1B}(L\theta) + P_{1B}(2L\theta) = M_{P}\theta + M_{P}(3\theta)$$

$$\boxed{@A}$$

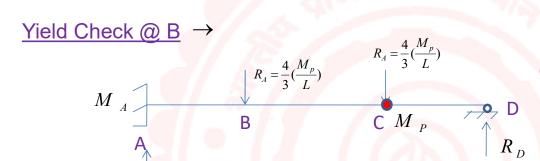
$$P_{U \ 1 \ B} = \frac{4}{3} \left( \frac{M_p}{L} \right)$$

This is lower than P'w Check for equilibrium & yield condition.

<1> ABC 
$$\rightarrow R_A$$
 CD  $\rightarrow R_D$  check vertical equilibrium.

<2> Point A(assume 
$$M_A \neq M_P$$
)
<2> Point B

## YIELD CHECK @ B



Equilibrium of CD and then overall vertical equilibrium

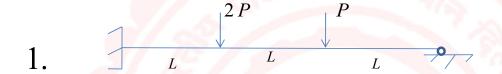
$$R_D = \frac{M_p}{L} \qquad R_A = \frac{5}{3} \left( \frac{M_p}{L} \right)$$

Overall moment equilibrium about A 
$$M_A = M_p$$

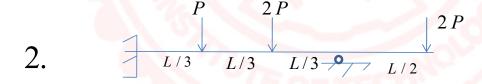
$$M_{B} = R_{A}L - M_{A}$$
$$= \frac{5}{3}M_{p} - M_{p}$$
$$M_{B} = \frac{2}{3}M_{p}$$

Yield check satisfied.

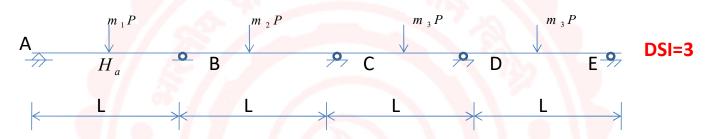
# PRACTICE PROBLEMS



### First solve by first principles



# APPLICATION OF SHORT-CUT APPROACH ON MULTISPAN STRUCTURE



### Possible independent mechanisms = 7 - 3 = 4



# APPLICATION OF SHORT-CUT APPROACH ON MULTISPAN STRUCTURE

#### **Mechanism 2**

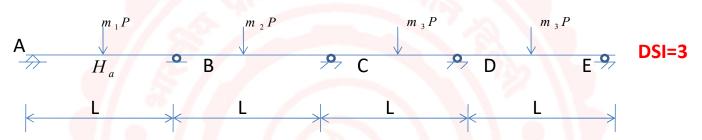


#### **Mechanism 3**





### **ALTERNATE FOUR MECHANISMS**

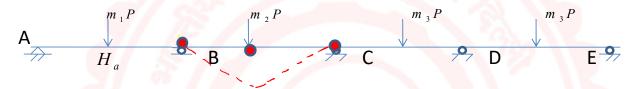


### Possible independent mechanisms = 7 - 3 = 4

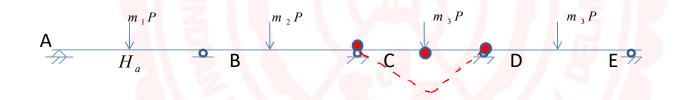


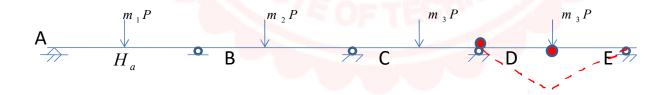
# APPLICATION OF SHORT-CUT APPROACH ON COMPLICATED STRUCTURE

#### **Mechanism 2**

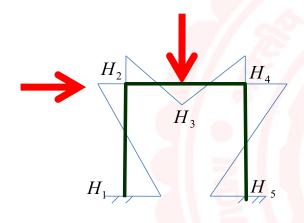


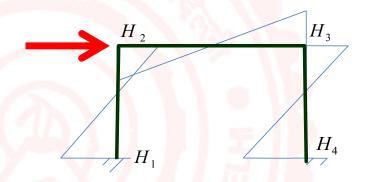
#### **Mechanism 3**





# 2 D FRAMES





$$m = n - r = 2$$

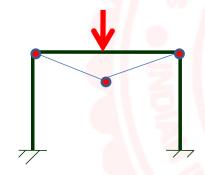
$$5 \quad \text{Sway}$$

$$m = n - r = 1$$

# 2 D FRAMES

### **Symmetric**

Under symmetrical vertical loads
No sway frame



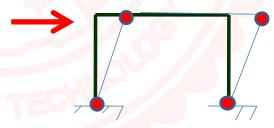
Beam mechanism (local collapse)

Under horizontal loads or unsymmetrical vertical loading or unsymmetrical structure under vertical loading Sway frame

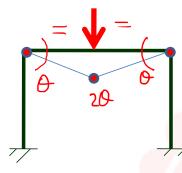
Beam mechanism

Sway mechanism.

Beam + Sway - Combined Mechanism

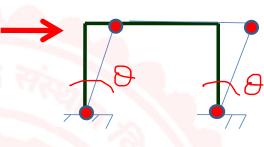


Sway mechanism (complete collapse)

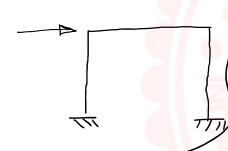


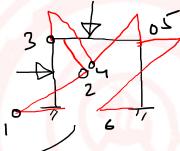
$$N = 6$$

$$T = 3$$

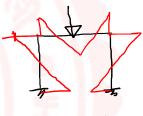


#### Beam mechanism



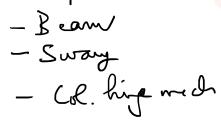


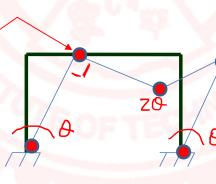
#### Sway mechanism

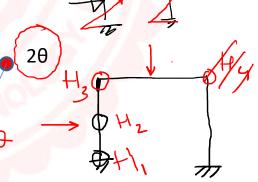


#### Combined mechanism

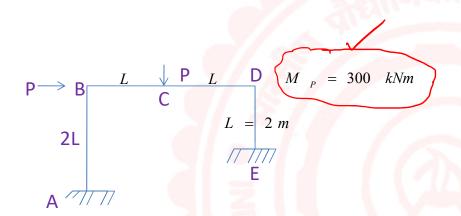








### **EXAMPLE**

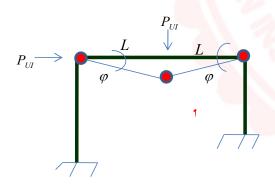


$$m = n - r = 2$$

Independent

- 1 Beam
- Sway
- 3 Combined.

#### <1> BEAM MECHANISM

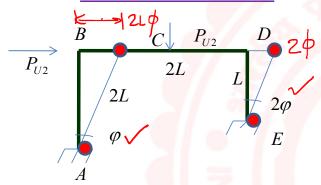


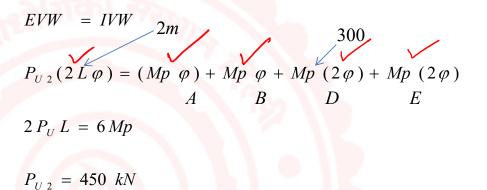
$$EVW = IVW$$

$$P_{UI} L \varphi = \underline{Mp \ \varphi} + \underline{Mp \ (2\varphi)} + \underline{Mp \ \varphi}$$

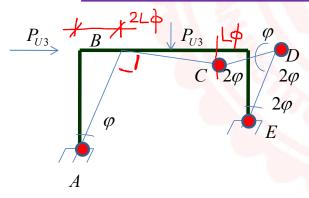
$$2m P_{UI} = 600 \ kN \ B$$

#### <2> SWAY MECHANISM





#### <3> COMBINED MECHANISM



No work done at "B" due to mutual cancellation....

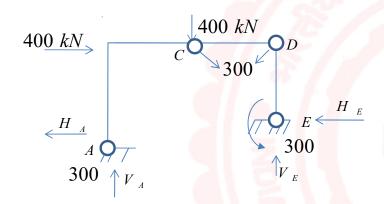
$$EVW = IVW$$

$$P_{U3}(2L\varphi) + P_{U3}(L\varphi) =$$

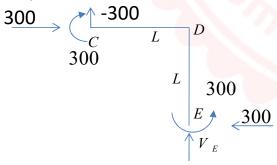
$$Mp \varphi + Mp (2\varphi) + Mp (3\varphi) + Mp (2\varphi)$$

$$P_{U3}(2L\varphi) + Mp$$

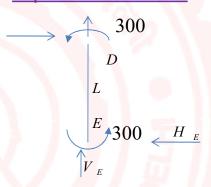
Combined mechanism is the critical (true) mechanism provided equilibrium & yield conditions are satisfied.



#### **Equilibrium of CDE** -

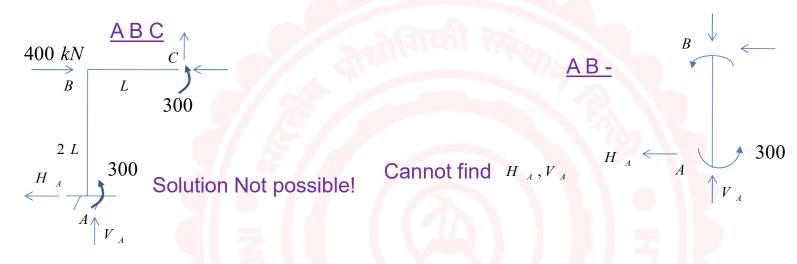


#### **Equilibrium of DE** -



$$H_{E} L = 300 + 300$$
  
 $H_{E} = 300 kN$ 

Moment about 
$$C$$
 – 300  $L$  =  $V_E$   $L$   $V_E$  = 300  $kN$ 

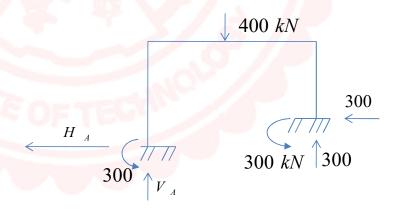


From horizontal /vertical equilibrium -

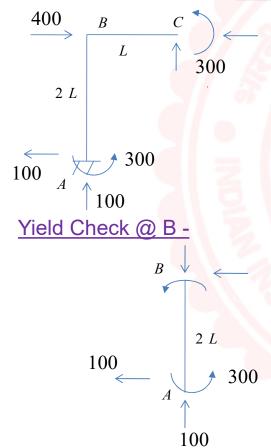
$$H_A = 100 kN$$

$$V_A = 100 kN$$

Use equilibrium of over all structure



#### Check equilibrium of ABC



#### Check moment equilibrium about C

$$M_{C} = 100 \times L + 100 \times 2L - 300 - 300$$
  
= 0

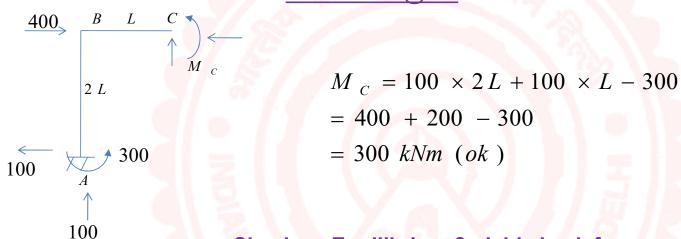
#### Moment equilibrium satsified.

$$M_{B} = 100 \times 2L - 300$$

$$= 100 \text{ kNm}$$

$$< M_{D} \text{ ok}$$

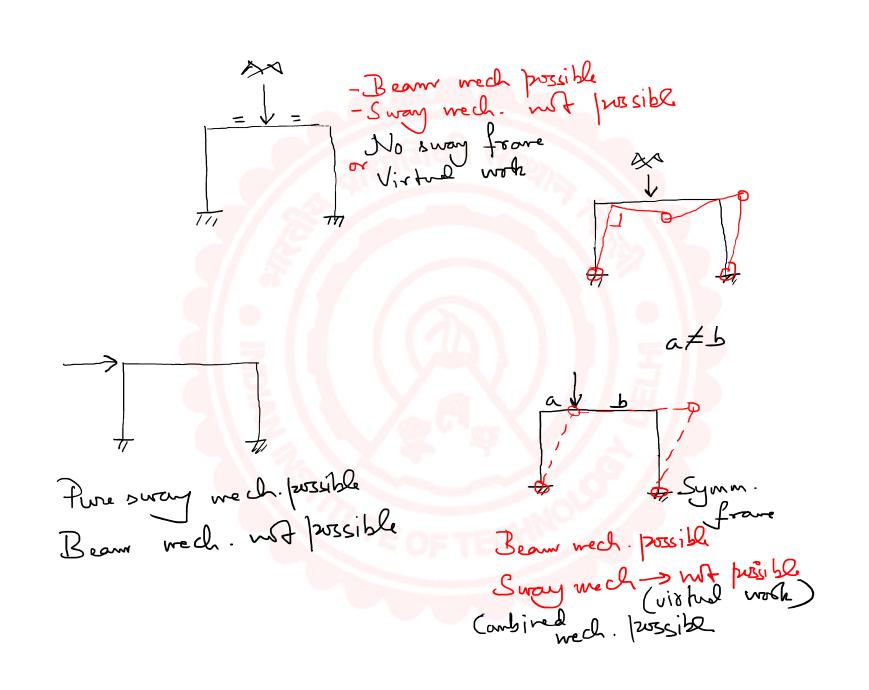
#### Yield Check @ C -



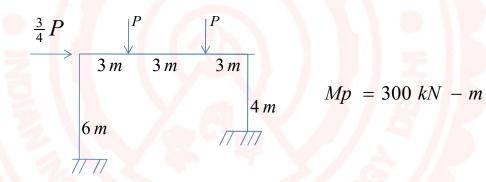
<u>Check:</u> Equilibrium & yield check for other mechanisms.

HOW DOES THE GOVERNING MECHANISM CHANGE WHEN THE STRUCTURE IS UNDER (A) ONLY VERTICAL LOADS (B) ONLY HORIZONTAL LOADS

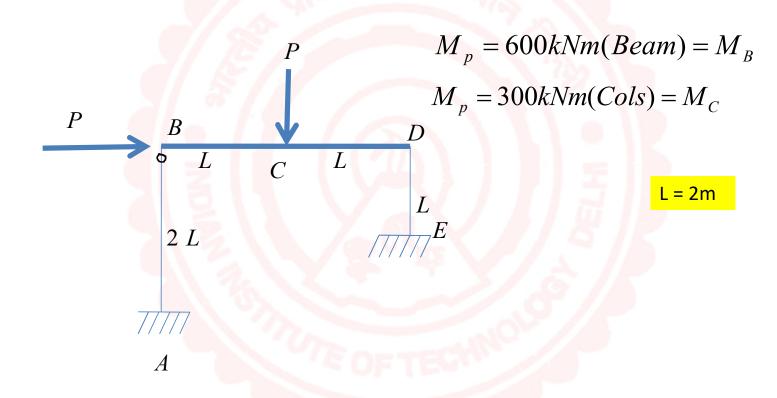
(A) Sway mechanism not possible (B) beam mechanism not possible



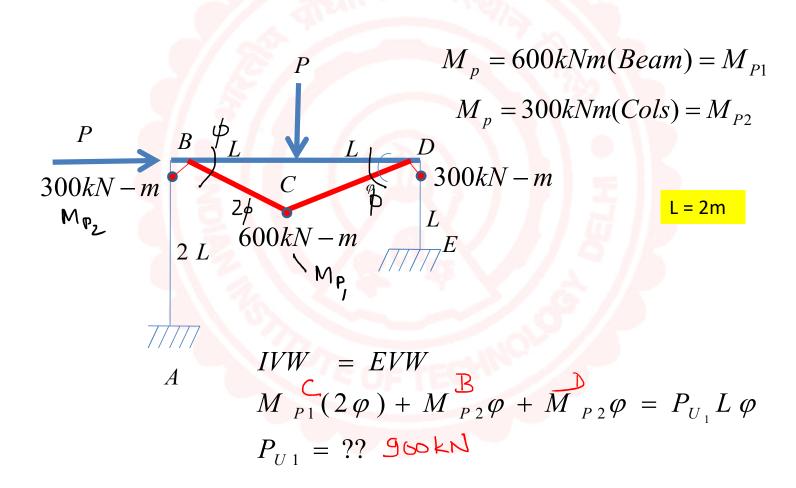
# PRACTICE PROBLEM



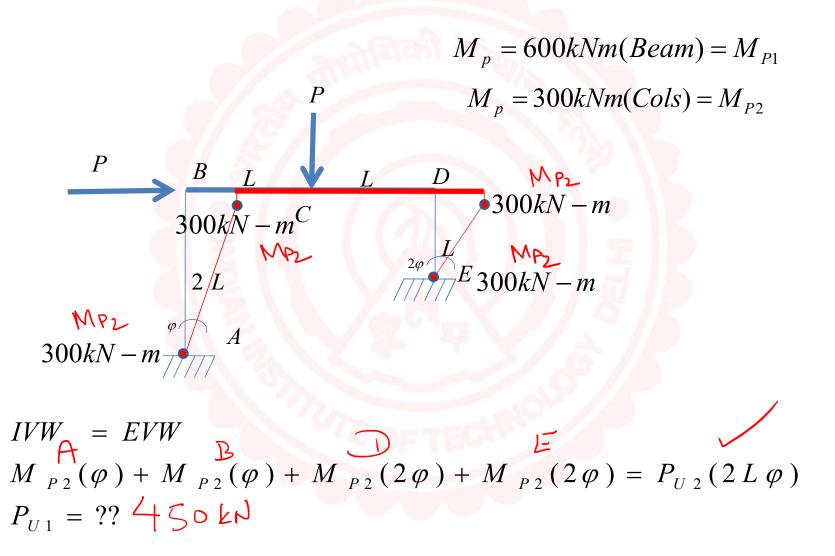
# **EXAMPLE**



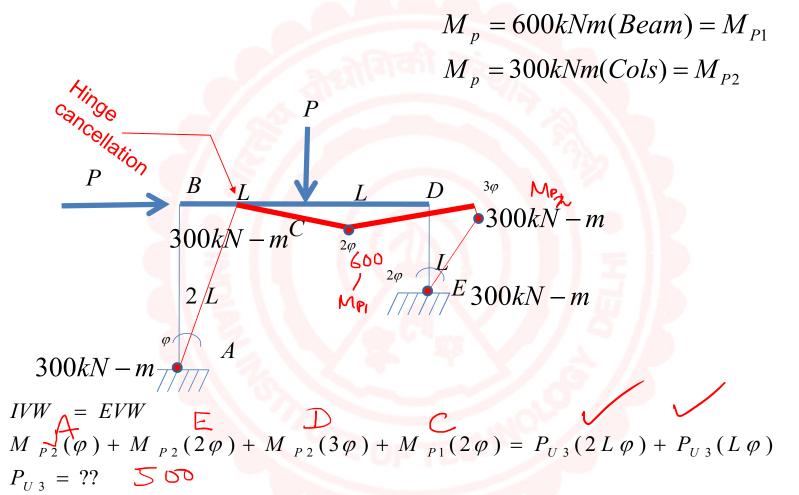
# Beam Mechanism



## Sway Mechanism

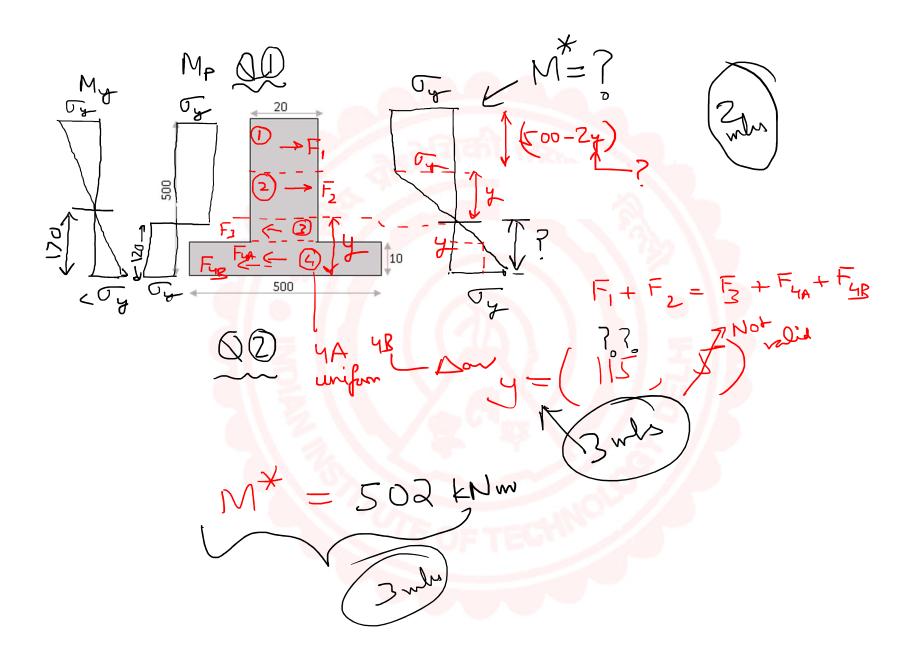


### **Combined Mechanism**

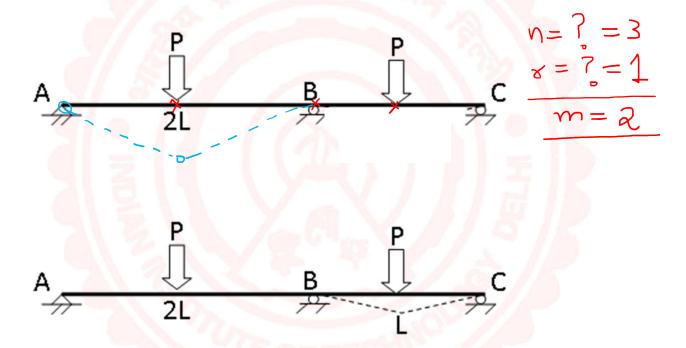


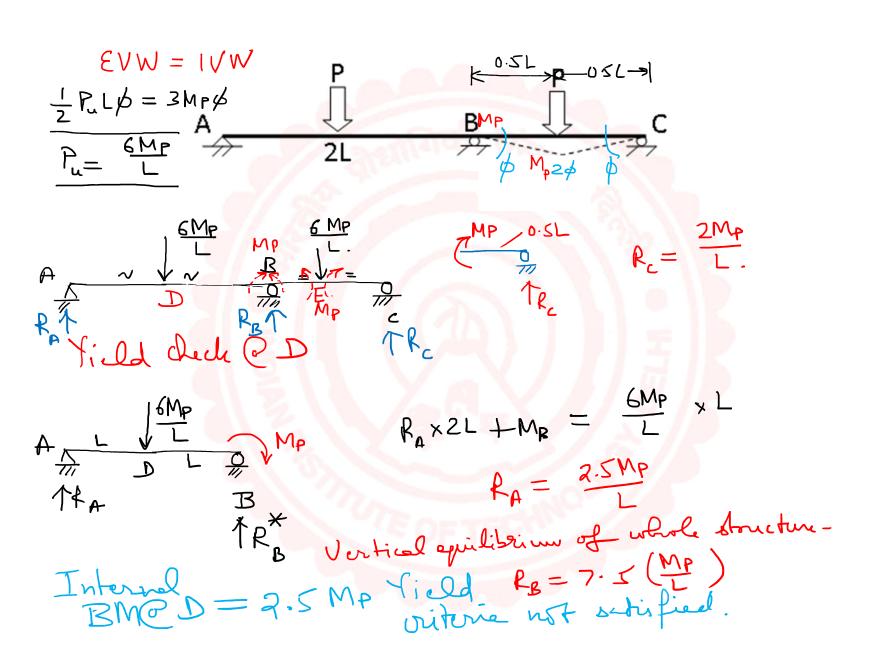
**CHECK FOR EQUILIBRIUM AND YIELD CRITERIA** 

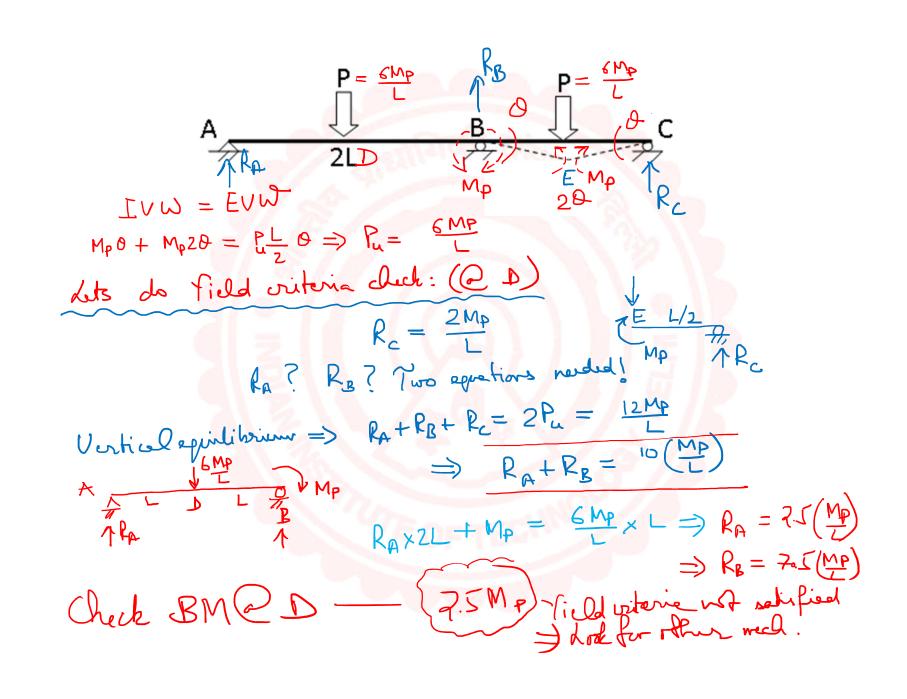
Only Rosizontal brads Beam Hinge concellation Possibility
of sway
wech. under
writelloods in

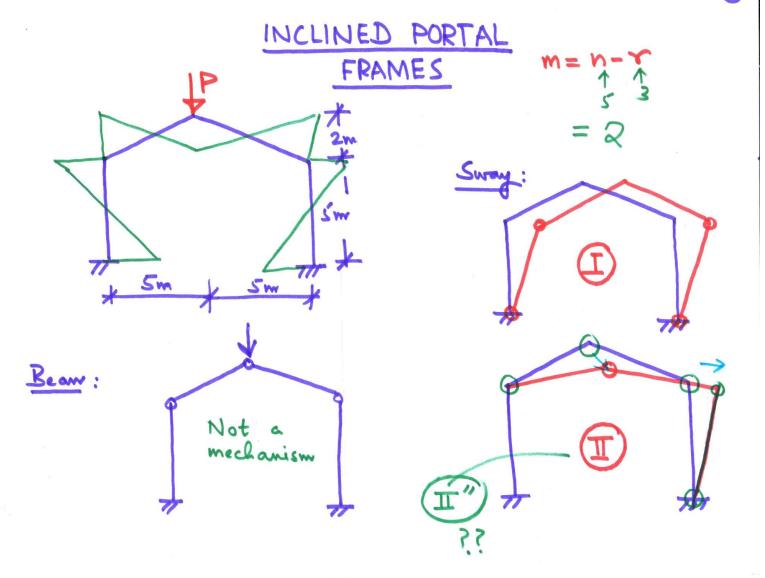


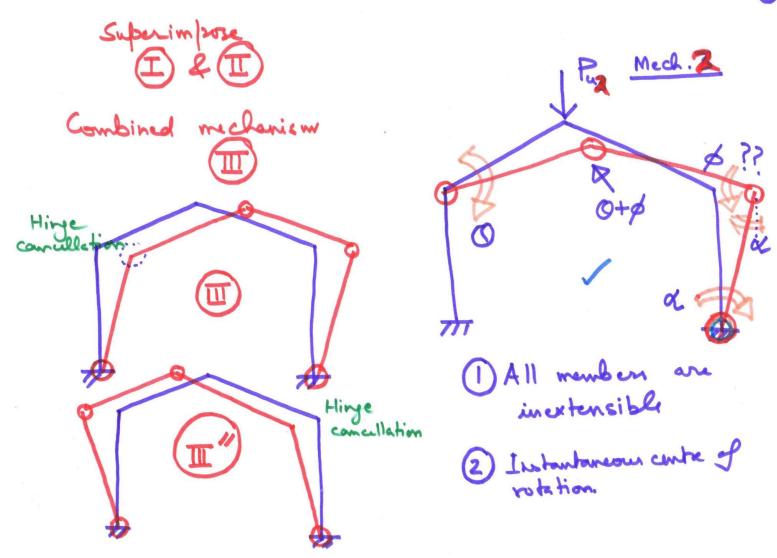
# EXAMPLE: To check for a mechanism to be critical

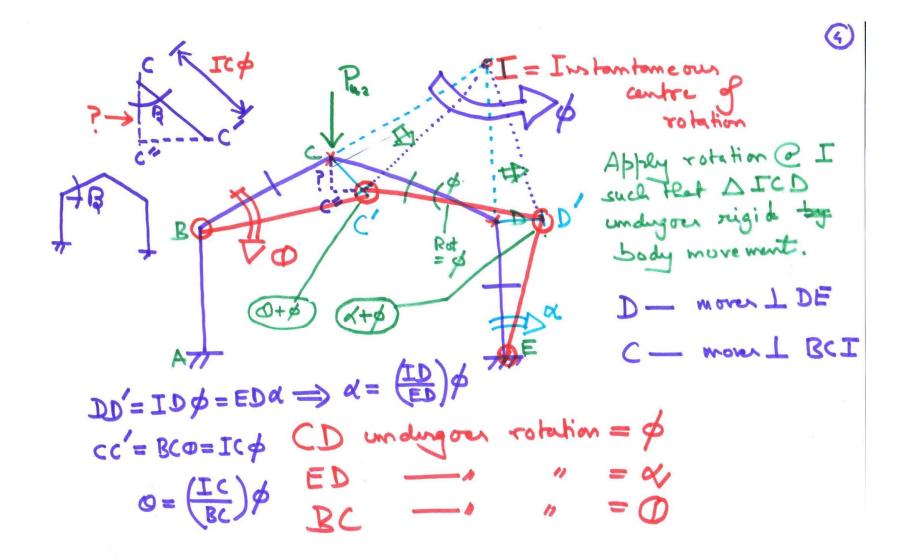






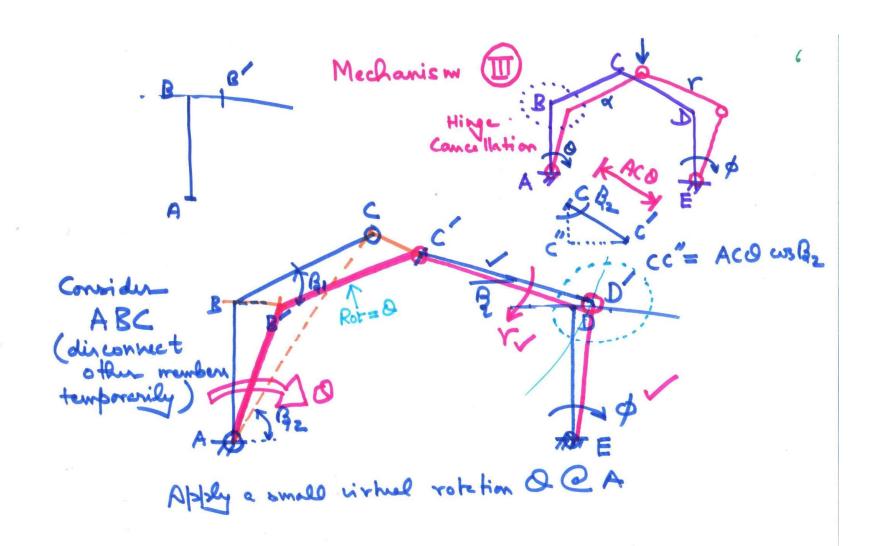


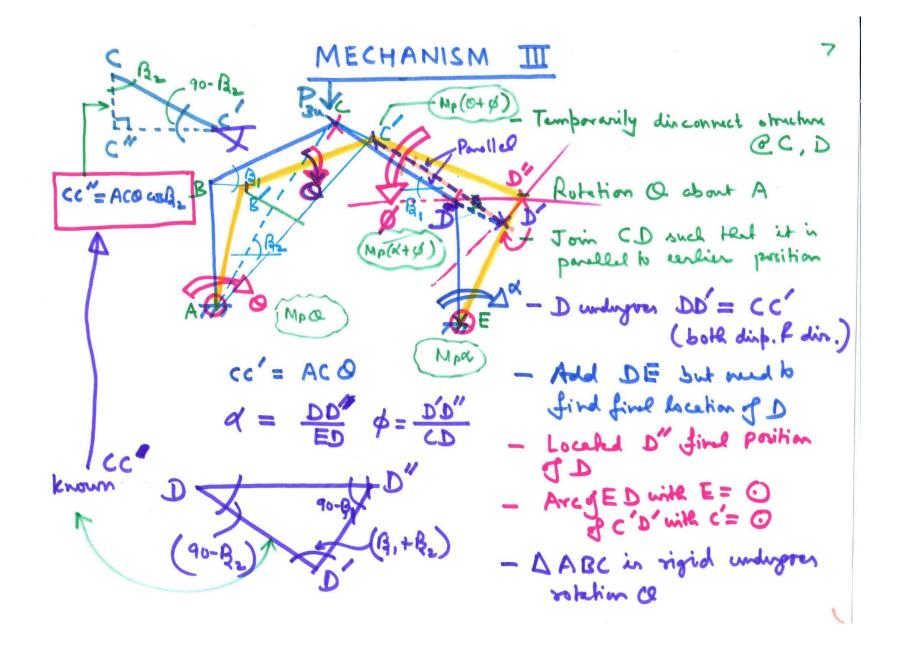




B
$$C = \frac{10}{5} \phi$$

$$C$$





$$\frac{DD''}{\sin(\theta_1 + \theta_2)} = \frac{DD''}{\cos(\theta_2)} = \frac{DD'}{\cos(\theta_2)}$$

$$\frac{EDd}{\sin(\theta_1 + \theta_2)} = \frac{CDcd}{\cos(\theta_2)} = \frac{AC^{(1)}\cos(\theta_2)}{\cos(\theta_1)}$$

$$d = \frac{AC^{(1)}\sin(\theta_1 + \theta_2)}{ED\cos(\theta_1)} = \frac{AC\cos(\theta_2)}{CD\cos(\theta_1)} = \frac{AC\cos(\theta_2)}{CD\cos(\theta_1)} = \frac{AC\cos(\theta_2)}{CD\cos(\theta_1)}$$

Solve and orbtain Puz = 1.5Mp

Upper bound them

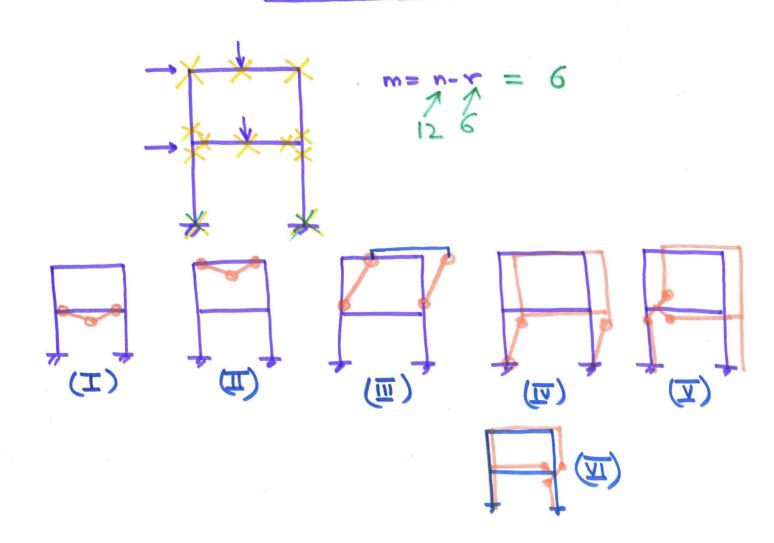
Pu = Smaller of (Puz & Puz)

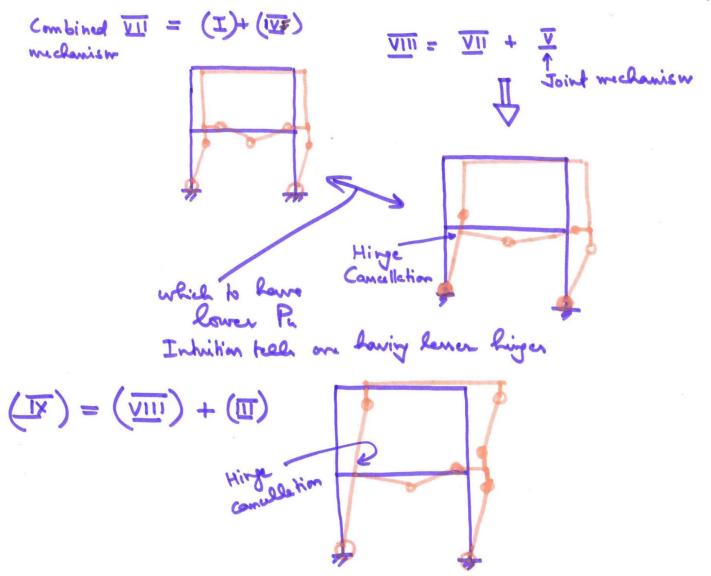
[eliminating away mech. I]

Equilibrium & yield click to be done to sule out any mechanism boung bean missed out.

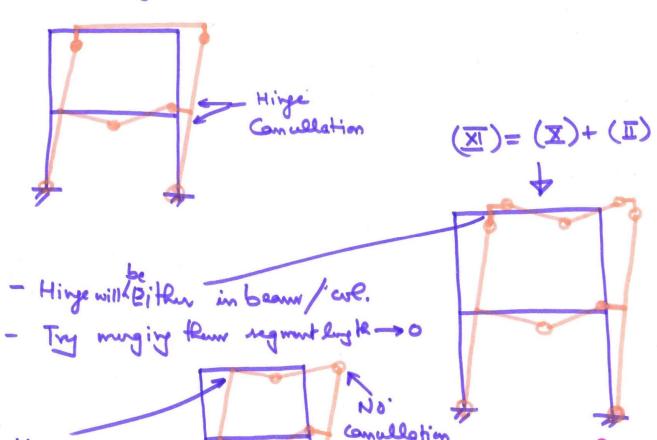
(H.W)

### TWO STOREY FRAME





$$(x) = (\underline{x}) + (\underline{x})$$



Hinge Concellation (Irrespective of strugtle of Seam or column)

